

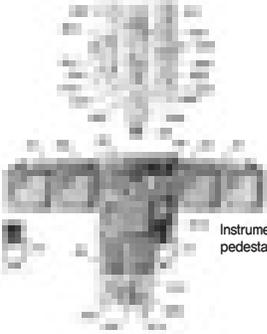
**Structure and general**

The main key refers to the MV-22 (US Marine Corps) aircraft. Items specific to the CV-22(US Air Force) are identified accordingly.

Typically the pilot sits on the right in the MV-22 and on the left in the CV-22.

- 1 Nose cone-composite
- 2 "Anti-plow"bulkhead-composite
- 3 Windscreen enclosure frame-titanium
- 4 Side console-both sides
- 5 Cockpit side window/emergency escape hatch with pyrotechnic jettison-co-pilot's side window removed for clarity
- 6 Standby magnetic compass
- 7 Instrument panel and shroud
- 8 Look-down window
- 9 Rear view mirror
- 10 Co-pilot's fully adjustable crashworthy stroking seat-armoured composite
- 11 Canted cockpit bulkhead-with seat mount
- 12 Seat occupant weight selector
- 13 Port side forward avionics bays
- 14 Skin panels-graphite-epoxy solid laminate with wire mesh lightning discharge membrane
- 15 Forward fuselage/centre fuselage interface
- 16 Upper fuselage strake-aluminium alloy with frangible leading edge
- 17 Crew chief location (aft of centre console)-MV-22
- 18 Starboard (stbd) side avionics bay
- 19 Cockpit roll-over protection beams
- 20 Two-part crew door-up-and-over section and lower fold-down section with integral steps
- 21 Upper door tracks
- 22 Cockpit door with troop commander's crashworthy stroking seat-facing into cabin
- 23 Upper deck avionics rack and support frames
- 24 Hand hold
- 25 Wing stow upper deck-composite
- 26 Wing stow fairing locators
- 27 Deck drain
- 28 Cavin compass
- 29 Life raft stowage-two rafts
- 30 Cabin underfloor side beam
- 31 Sponson-graphite-epoxy solid laminate with wire mesh lightning discharge membrane
- 32 Tie-down shackles
- 33 Sponson/main landing gear forward bulkhead-machined aluminium alloy
- 34 Tie-down net stowage
- 35 Foldaway crashworthy troop seats-during evacuation seats located below escape hatches are removed via a pyrotechnic actuator
- 36 Wing stow/fuselage frame-aluminium alloy
- 37 Fixed cabin floor panels with cargo tie-downs-access to cargo hooks is via two hinged panels
- 38 Cargo handling roller conveyors-removable
- 39 Cabin escape hatch-three off
- 40 Cabin 1.82x1.82x7.31m-capable of accommodating 24 combat loaded marines (fast ropes and parachute static lines are provided), 12 litters or 9,000kg (20,000lb) of internal cargo. External cargo is carried by single and dual cargo hooks-4,500kg (10,000lb) and 6,800kg (15,000lb) capability located beneath the cabin floor
- 41 Overwing fairing panel-composite
- 42 Conversion actuator forward wing panel
- 43 Wingtip/spindle rib-aluminium alloy
- 44 Air intake/proprotor gearbox fairing
- 45 Spinner avionics shelf
- 46 Nacelle upper conversion joint fairing
- 47 Trailing edge cove support structure-composite
- 48 Flaperon hinge mount-aluminium alloy
- 49 One piece upper and lower wing skins with bonded stringers-composite
- 50 Leading edge vortex generators
- 51 Wing/flex ring rear mount and lifting point
- 52 Centre wing vortex generators
- 53 Upper wing skin fairings
- 54 Auxiliary power unit "A"frame mount
- 55 Mid-wing gearbox support beam
- 56 Upper deck avionics rack attachment points
- 57 Wing/flex ring forward mount and lifting point
- 58 Centre wing dry bay
- 59 Mid-wing gearbox central mount
- 60 Controlled wing failure zone-designed to separate from cabin in the event of a crash
- 61 Mid-wing gearbox diagonal mount
- 62 Lower wing skin inspection panels
- 63 Wing rib-built-up composite
- 64 Wing forward and rear spars-composite
- 65 CV-22 wing fuel cell locations-plumbing fitted as standard on all models
- 66 Main landing gear support beam-machined aluminium alloy
- 67 Sponson strake
- 68 Spindle inboard rib-machined aluminium alloy
- 69 Nacelle maintenance platforms
- 70 Nacelle lower access panel
- 71 Nacelle primary structure-mixed construction of titanium/aluminium alloy and composite
- 72 Primary fire wall-titanium
- 73 Oil cooler intake
- 74 Wing fence
- 75 Maintenance step
- 76 Sponson/main landing gear rear bulkhead-machined aluminium alloy
- 77 Overwing aft fairing
- 78 Mid-wing gearbox access door
- 79 Cabin insulation blankets-fire proof
- 80 Sponson aft bulkhead-machined aluminium alloy
- 81 Cargo loading ramp-multi-position

- 82 Fold out ramp extensions
- 83 Ramp actuator fitting
- 84 Centre fuselage/rear fuselage interface
- 85 Rear fuselage overhead escape hatch
- 86 Rear fuselage frames-composite
- 87 Flapper door upper clearance panels-pneumatic
- 88 Flapper door
- 89 Ramp longeron and fairing-composite
- 90 Rear fuselage torque box-both sides
- 91 Rear fuselage/tail attachment interface-aluminium alloy
- 92 Empennage forward attachment points
- 93 Empennage leading edge-composite
- 94 Two spar single piece tail plane-mixed construction
- 95 Elevator actuator access panel
- 96 Fin attachment points
- 97 Two spar single piece fin-composite
- 98 Fin leading edge-detachable
- 99 Handhold/fire extinguisher nozzle access
- 100 Windscreen washer bottle



**Instrument panel, overhead panel and centre pedestal detail**

**Environmental control system(ECS)and anti-ice**

The ECS provides a positive pressure differential up to 0.062 bar (0.9lb/in<sup>2</sup>)for the cockpit, and 0.048 bar (0.7lb/in<sup>2</sup>)for the cabin in possible nuclear/biological/chemical(NBC)environments.

- A1 Electrically heated windscreen panels
- A2 Side window demister ducts
- A3 Cockpit pressurisation safety valve
- A4 Overhead gaspers
- A5 Cockpit heating duct
- A6 Avionics bay forward cooling system
- A7 Avionics bay aft cooling system
- A8 Avionics bay cooling system filter
- A9 Cockpit outflow valve
- A10 Avionics cooling air inlet duct to A8
- A11 Inlet air particle separator exhaust and drain
- A12 Avionics cooling air inlet particle separator
- A13 Avionics bay exhaust fan and outlet
- A14 Onboard oxygen generating system(OBOGS) and onboard inert gas generating system(OBIGGS) concentrators
- A15 Cabin air distribution duct
- A16 Cabin air crossover ducting
- A17 Compressed air duct from the shaft driven compressor (SDC)
- A18 Environmental control unit(ECU)cooling ram air inlet
- A19 ECU heat exchanger
- A20 ECU exhaust
- A21 ECU ground pneumatic supply connector
- A22 ECU emergency air vent fan
- A23 NBC filter pack
- A24 ECS controller
- A25 SDC inlet

- A26 SDC inlet particle separator
- A27 SDC
- A28 Leading edge electrically controlled pneumatic actuated de-icing boots (SDC supplied air via the ECU)-three per wing
- A29 Proprotor blade spinner and side panel anti-ice system-structural wire mesh electrically heated
- A30 Emergency oxygen bottle
- A31 De-icing boot air supply
- A32 SDC inlet screen
- A33 De-ice valves

**Flying controls**

- Flying controls operate on a triple redundant fly by wire system (operating within an automatic flight envelope protection system)-all surfaces are hydraulically controlled
- C1 Co-pilot's rudder pedals and toe brakes
  - C2 Co-pilot's side console mounted thrust control lever (the pilot's thrust control lever is located on the centre pedestal)-both incorporate nacelle control and electronic warfare (EW)controls
  - C3 Co-pilot's cyclic control lever-incorporating trim, communication, multifunction display (MFD), helmet mounted display (HMD)and cargo release controls
  - C4 Cyclic control pitch trim actuator
  - C5 Spindle/blade grip adaptor fork
  - C6 Spindle

- C7 Pendulum damper weight assembly and elastomeric spring
- C8 Glass reinforced fibre yoke, elastomeric thrust bearing and beam assembly
- C9 Pendulum damper yoke
- C10 Constant velocity joint
- C11 Proprotor blade grip and pitch horn
- C12 Root Fairing
- C13 Proprotor blade "fangs" - fold point

**Nacelle conversion detail**

- C14 Cordwise dynamic balance weights
- C15 Erosion strip
- C16 Proprotor blade(designed to "broomstraw"fall upon ground contact)-graphite and glass reinforced fibre spar and skin with honeycomb filled after body
- C17 Spanwise dynamic balance weights
- C18 Nacelle conversion actuator backup drive-hydraulic
- C19 Conversion actuator upper section
- C20 Nacelle dual primary conversion actuator-hydraulic

**Infrared suppressor detail**

- C21 Primary conversion actuator spindle and bearing
- C22 Conversion actuator lower section(shown in lowered position)-upper actuator section retracts into lower section
- C23 Single slotted outboard flaperon-composite construction
- C24 Flaperon sealing shroud
- C25 Shroud cam track and guide
- C26 Flaperon hydraulic actuator-two per flaperon
- C27 Inboard flaperon
- C28 Remote hover control-used from the hoist position
- C29 Flaperon uppermost deflection
- C30 Flaperon lowermost deflection
- C31 Non-rotating swashplate anti-drive assembly

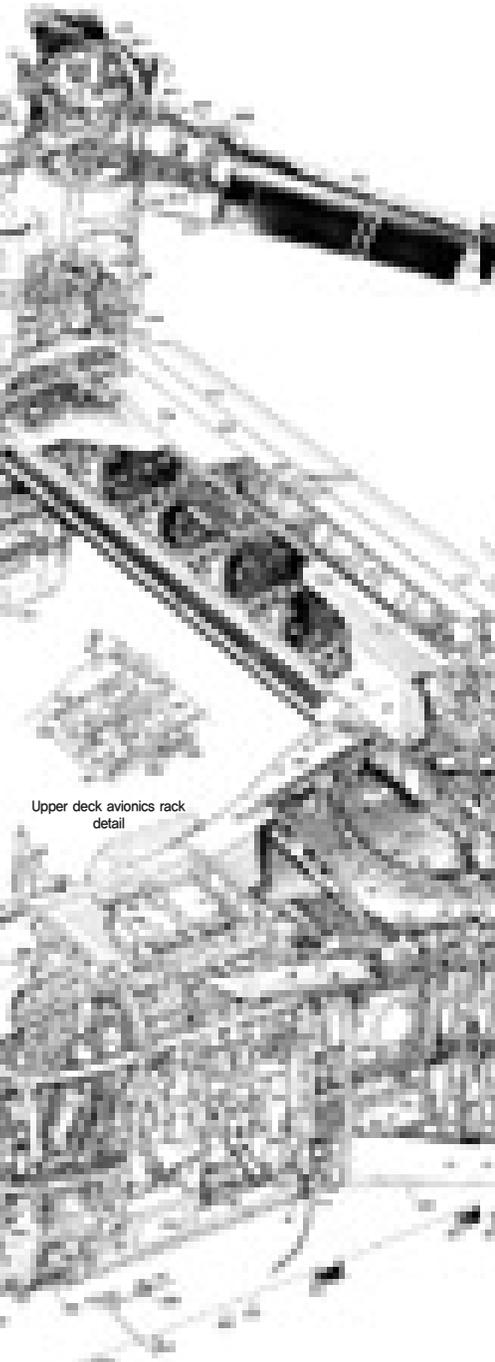
**Avionics bays detail**

- C32 Pitch links
- C33 Swashplate-rotating
- C34 Swashplate-non-rotating
- C35 Swashplate hydraulic actuators-three off
- C36 Rudder hydraulic actuator
- C37 Rudder-composite structure
- C38 Single piece elevator-composite structure
- C39 Elevator hydraulic actuator-three off
- C40 Elevator mounted hinge-five off
- C41 Hinge mount
- C42 Hinge link
- C43 Rotor brake air inlet
- C44 Flaperon hinge

- D1 Full colour multifunction primary flight displays
- D2 Control display unit(CDU)
- D3 Standby flight instrument display
- D4 Standby flight instruments
- D5 Flight control ppanel-plus(CV-22)multi-mode radar control
- D6 Remote frequency indicator
- D7 Master alert panel

**Instrument panel**

- D1 Full colour multifunction primary flight displays
- D2 Control display unit(CDU)
- D3 Standby flight instrument display
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- D5 Flight control ppanel-plus(CV-22)multi-mode radar control
- D6 Remote frequency indicator
- D7 Master alert panel



**Upper deck avionics rack detail**

- D8 Magnetic compass
- D9 CDU control panels
- D10 Landing gear
- D11 FLIR multifunction tracking device
- D12 Flap controls
- D13 Dedicated EW display
- D14 Parking brake lever
- D15 Nacelle control disable switches
- D16 Battery control
- D17 Cargo hook/hoist control
- D18 Cargo loading ramp
- D19 Proprotor blade fold and wing stow control
- D20 Lighting controls
- D21 AFU control
- D22 Flight controls
- D23 Engine controls-FADEC
- D24 Fuel dump control
- D25 Power steering controls
- D26 Fire Suppression
- D28 Emergency oxygen
- D29 VHF radio
- D30 Panel light

**Avionics and electrical**

- E1 AN/APQ 174D multi-mode radar-CV-22
- E2 Radar interface unit-CV-22

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- E3 In-flight refuelling probe light-white/infrared
- E4 Radar warning receivers
- E5 Forward looking infrared (FLIR) turret-AN/AAQ-27
- E6 Suite of integrated radio frequency counter-measures (SIRFC) forward receiver and tailing (both sides)-CV-22
- E7 Missile warning forward optical sensor (both sides)-AAR-47
- E8 Radar sensor (both sides)-ARP-39A
- E9 Windscreen wipers-two off
- E10 Pitot static probes-1 and 3 on port side 2 and 4 on stbd side
- E11 Angle of attack vane
- E12 Push to talk foot switch
- E13 FLIR multifunction tracking device control
- E14 Glide slope antenna
- E15 TACAN antenna
- E16 IFF antenna-upper

- E17 Helmet mounted display (HMD) connector
- E18 SIRFC detect band antenna
- E19 Cockpit emergency escape hatch external initiator
- E20 SIRFC antenna-CV-22
- E21 Troop commander antenna
- E22 Radar altimeter receiver
- E23 Radar altimeter transmitter
- E24 SIRFC forward transmitter (both sides)-CV-22
- E25 Radio No 4 line of site (LOS) antenna
- E26 SIRFC transmitter and receiver bay-CV-22
- E27 Ground power receptacle
- E28 Cabin emergency escape hatch external initiator
- E29 AN/AVR-2A sensor (both sides)
- E30 Cabin emergency escape hatch internal initiator handle
- E31 Troop seat pyrotechnic actuator release

- E59 Radar sensor (both sides)-ARP-39A
- E60 AN/ALE-47 Chaff/flare dispenser-MV-22
- E61 Interference canceller-CV-22
- E62 Chaff/flare sequencer
- E63 SIRFC receiver amplifier-both sides-CV-22
- E64 Utilities panel-electrical

- E84 Cabin lighting control
- E85 No 1 circuit breaker panel-four off throughout aircraft
- E86 Overhead panel
- E87 Avionics bays equipment distribution
- E88 Hoist floodlight-white/infrared
- E89 Battery (cabin ceiling mounted below mid-wing gearbox)-15amp-hour
- E90 SIRFC cooling fan-CV-22
- E91 Communications panel
- E92 Pivoting landing light-white/infrared
- E93 Gear-down light-white
- E94 Anti-collision strobe-red

#### Fuel system

Total fuel capacity-4,344 litres (1,149 USgal). Including two internally mounted ferry tanks-11,541 litres

- P18 Transmissin adpater
- P19 TAGB accessories pad and integral oil reservoir-drives oil filters, oil pumps hydraulic pumps/stbd side-hydraulic system No 2 and port side-hydraulic system No 1 and electrical generators

- P20 Pylon spindle
- P21 Oil cooler
- P22 Rear nacelle blower assembly
- P23 Ram air inlet-infrared suppressor
- P24 Infrared suppressor ram air inlet ducting
- P25 Infrared suppressor mixer side plenum panels
- P26 Interconnect drive shafts (five per wing) - interconnected via the mid-wing gear box (MWGB) and the TAGB-distribute power to both PRGB's in the event of single engine shutdown

- P27 Oil cooler exhaust
- P28 Oil cooler blower
- P29 Oil cooler
- P30 Hamilton Sundstrand T-62T-46-2 auxiliary power unit (APU)-rated at 224KW(300shp)

- P31 APU exhaust
- P32 Mid-wing gearbox
- P33 Rotor phasing unit
- P34 Double coupling and support assembly
- P35 Engine exhaust transition duct-IR suppressor
- P36 IR ram air centre body mixer
- P37 Coanda deflector tubes
- P38 Emergency PRGB oil lubrication reservoir and pump
- P39 Single coupling and support assembly
- P40 Fire suppression gas generators-seven in midwing area, three in each wing trailing edge cove, one on each tip rib and one in each wing (located inboard of wing fuel cells)

#### Undercarriage and hydraulics

The triple redundant hydraulic system typically operates at 345 bar (5,000lb/in<sup>2</sup>)

- U1 Rearward retracting oleo pneumatic nose landing gear (NLG)-electronically controlled, hydraulically actuated and steered-Messier Dowty
- U2 NLG doors-composite
- U3 Access to foot brake pilot master cylinders-both side
- U4 Parking brake transfer valves
- U5 Cabin internal cargo loding winch-hydraulically actuated

- U6 Ressue hoist-hydraulically actuated
- U7 Main landing gear (MLG) tyre door-composite
- U8 MLG closing panel-composite
- U9 MLG Pivot
- U10 MLG drag strut
- U11 MLG retraction actuator
- U12 Forward retracting oleo pneumatic twin wheel MLG-electronically controlled, hydraulically actuation with dual hydraulic multi-disk carbon braking system-Messier-Dowty

- U13 Weight-on-wheel switch
- U14 MLG outboard door-composite
- U15 MLG oleo upper access door
- U16 Hydraulic fluid level monitor
- U17 Hydraulic reservoir servicing ports and filters
- U18 Cargo loading ramp actuator-two off
- U19 emergency pneumatic reservoir and electrical pump (stbd side rear sponson)-landing gear and cargo loading ramp deployment
- U20 Hydraulic pump No 3-No 1 the port TSGB pad and No 2 on the stbd TAGB pad

- U21 Hydraulics service point
- U22 Fipper door actuators-hydraulic
- U23 Cargo loading ramp primary locks-hydraulic
- U24 Hoist winch control
- U25 Hike extend/vents

#### Wing storage and blade fold

Fully automated prop rotor blade fold and wing storage is accomplished in 90s in 45KT(89m/h) winds

- W1 Wing storage deck supports
- W2 Wing locking flag type indicator (black-locked, chequered-unlocked)-one per lockpin actuator
- W3 "Flexing" secondary locator fittings/guides with lead running strip lubrication-eight off
- W4 Wing storage a "flexing"-stainless steel
- W5 Lockpin primary locators-four off attachment points-four off
- W6 "Flexing" to wing integrally machined attachment points-four off
- W7 Wing storage lockpin control module-hydraulic
- W8 Lockpin actuator (hydraulic)-four off
- W9 Wing storage capstan drive with parallel action cables

- W10 Wing rotation control-hydraulic
- W11 Storage cables-two off
- W12 Wing storage manual drive shaft
- W13 Wing storage "cap stand"-systems rotational interface between the wing and fuselage
- W14 Prop rotor blade pitch locking pin
- W15 Prop rotor blade latch and sensor
- W16 Prop rotor blade fold drive unit incorporating electrical motor, "planetary" gear sets and brake mechanism
- W17 Prop rotor manual fold drive
- W18 Prop rotor blade fold control unit - electrical

(3,049 USgal), all fuel cells are nitrogen inert and have strategic self-sealing areas. CV-22 fuel capacities are 7,710 litres (2,037 USgal) and 13,131 litres (3,469 USgal) respectively.

- F1 In-flight refuelling probe
- F2 Sponson forward bladder-type fuel cell-both sides
- F3 Gravity filler point
- F4 Sponson fuel cell access
- F5 Sponson fuel cell climb/dive valve
- F6 OBIGGS unit
- F7 Single point refuel (SPR)/detuel connector
- F8 SPR panel
- F9 Sponson aft bladder-type fuel cell-stbd side only
- F10 Wing auxiliary bladder-type fuel cell (No 4)-CV-22
- F11 Wing auxiliary bladder-type fuel cell (No 3)-CV-22
- F12 Wing auxiliary bladder-type fuel cell (No 2)-CV-22
- F13 Wing auxiliary bladder-type fuel cell (No 1)-CV-22
- F14 Fuel vent line-CV-22
- F15 Fuel vent surge trap-CV-22
- F16 Feed cell pump
- F17 Bladder-type fuel feed cell-both wings
- F18 Hover in-flight refuelling connector and filter-located in door entrance
- F19 Fuel cell through flow tubes
- F20 Fuel inlet port
- F21 Fuel feed line
- F22 Fuel feed tank vent and surge traps

#### Powerplant and transmission

- P1 Prop rotor mast
- P2 Oil filter
- P3 Prop rotor gear box (PRGB)
- P4 PRGB integral oil reservoir
- P5 Pylon drive shaft interface
- P6 Torque housing
- P7 Engine air intake
- P8 Engine air particle separator (EAPS) splitter vane
- P9 EAPS-twin hydraulically powered blowers
- P10 EAPS exhaust
- P11 Engine fire extinguisher-one per engine
- P12 Pylon drive shaft and couplings
- P13 Engine starter-hydraulic
- P14 Rolls-Royce/Allison T406-AD-400 tuboshaft engine (FADEC controlled)-rated at 4,580 kW(6,150shp)
- P15 Tilt axis gearbox (TAGB)
- P16 Pylon assembly-complete
- P17 Pylon support

- E65 SIRFC receiver processor-CV-22
- E66 Overhead light
- E67 Rear fuselage overhead emergency escape hatch internal initiator handle
- E68 UHF/VHF antenna
- E69 Crash position indicator (CPI) antenna-CV-22
- E70 CPI-CV-22
- E71 Emergency egress lighting system-fitted to all cabin escape exits
- E72 WIA panel
- E73 SATCOM amplifier
- E74 SATCOM antenna
- E75 Multi-mission advanced tactical terminal antenna-CV-22
- E76 SIRFC modulator receiver-CV-22
- E77 SIRFC radio frequency switch-CV-22
- E78 WIA
- E79 FM homing antenna
- E80 Colling fan and exhaust
- E81 ADF antenna
- E82 Tail position light-red/infrared
- E83 Anti-collision light

Giuseppe Picarella  
Amarillo and Philadelphia  
2000

CV-22 aft  
sponson equipment

Nacelle/transmission/centre  
wing systems detail