A Design Process for Structural Borne Noise using Panel Contribution and Design Sensitivity

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Key Words: Structure-borne Noise(Analysis(), Noise Transfer Design Sensitivity Analysis()		Path Analy), S Contribution	Structural-Acous Analysis(stic Coupling),
In this study, we propose a more systematic d of 4 steps: Problem definition, Cause analysis improved the second step: Cause analysis. Accord of the panels of which panel contribution is position however, met the case in which the concept of phenomenon, we compared the major panels sele After investigating the difference between the two the second step in the design process consists of also DSA. It is finally validated that the propose transfer function more than 3.5 dB.	s, Developreding to the laive and large PCA is no exceed by PCA or results, and not only the	ss for the str ment of cou PCA(Panel Cer, results in valid in a A with the comore improve previous w	unter-mea Contribut a reduct few veh one chose yed proce yay: PCA	asure and vition Analysition in structicle tests. In the property of the prope	Validation. Es (s), a reduction ture-borne noi n order to und (Design Sensitiv sted. The prop mation analys	pecially, we in vibration se. We have, derstand this vity Analysis). osed one for is results but
1.						
				(Acoustic	cavity)	
20Hz ~ 200Hz			,	-	(Structura	l-Acoustically
(structure-borne noise)		Coupled s	ystem)			
가			•			
. Kim ⁽¹⁾						A: Transfer
	4	Path Anal	lysis)		(3)	
: (Problem definition),			•		(1.4)	
(Cause analysis), (Developm	nent of	-		leformation	_	
counter-measure) (Validation) Y K Zhang (2)	•				analysis) ^(1,2,5) ,	
, 1. 1x. Zhung		(Sensitivit	y analys	S1S)(0,7)		
(hybrid modeling) 5: (Root cause analysis),						
(Structure vibration measurement),						
(Panel acoustic contribution analysis),		•				
•	erimental					
verification) .	Timental				(Macc	damper)
				(damping		damper)
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NVH

Kim (1)

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2.

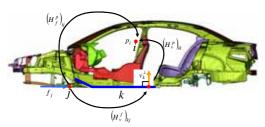


Fig. 1 Pictorial representation of transfer functions

(sound pressure) p (acoustic 가 (noise cavity) 가 transfer function) \mathbf{H}_{f}^{p} $\mathbf{p} = \mathbf{H}_f^p \mathbf{f}_s$ (1) $\Leftrightarrow p_i = \sum_{i=1}^{N_l} \left(H_f^p \right)_{ij} f_j$ $p_i = p(A, \varphi), (H_f^p)_{ij} = H(A, \varphi), f_j = f(A, \varphi)$ $\mathbf{p} = [\begin{array}{cccc} p_1 & \cdots & p_i & \cdots & p_{Nr} \end{array}]^{\mathrm{T}}, \qquad \mathbf{p} \in R^{Nr}$ $\mathbf{f} = [\begin{array}{cccc} f_1 & \cdots & f_j & \cdots & f_{Ni} \end{array}]^{\mathrm{T}}, \qquad \mathbf{f} \in R^{Ni}$ $\mathbf{H}_{f}^{p} = \left[\left(H_{f}^{p} \right)_{ij} \right] ,$ $\mathbf{p}, \mathbf{f}_s \quad \mathbf{H}_f^p \qquad p_i, f_{si} \quad \left(H_f^p\right)_{ii}$ \boldsymbol{A}

Fig. 1 .

2.1 (PCA: Panel contribution analysis)⁽²⁾

Fig. 2

v p

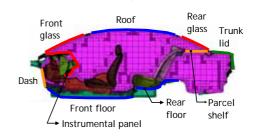


Fig. 2 Acoustic boundary and major panels

$$\mathbf{p} = \mathbf{H}_{v}^{p} \mathbf{v}^{\perp}$$

$$\Leftrightarrow p = \sum_{k=1}^{Np} p_{k} = \sum_{k=1}^{Np} \left(\mathbf{H}_{v}^{p} \right)_{k} v_{k}^{\perp}$$
(2)

where

Transfer Function)

where
$$p_{k} = p(A, \varphi), \quad (H_{v}^{p})_{ik} = H(A, \varphi), \quad v_{k}^{\perp} = v(A, \varphi)$$

$$\mathbf{v}^{\perp} = \begin{bmatrix} v_{1}^{\perp} & \cdots & v_{k}^{\perp} & \cdots & v_{Np}^{\perp} \end{bmatrix}^{T},$$

$$\mathbf{v}^{\perp} = \mathbf{H}_{f}^{v} \mathbf{f} \iff v_{k}^{\perp} = \sum_{i=1}^{Nj} \left(H_{f}^{v} \right)_{kj} f_{j},$$

$$\begin{split} \mathbf{H}_{v}^{p} &= \left[\begin{array}{c} \left(\boldsymbol{H}_{v}^{p} \right)_{k} \end{array} \right] \text{ and } \mathbf{H}_{f}^{v} = \left[\begin{array}{c} \left(\boldsymbol{H}_{f}^{v} \right)_{kj} \end{array} \right] \\ \mathbf{v}^{\perp} \quad , \quad \mathbf{H}_{v}^{p} \qquad \mathbf{H}_{f}^{v} \qquad \qquad v_{k}^{\perp} \; , \quad \left(\boldsymbol{H}_{v}^{p} \right)_{ik} \end{split}$$

$$(H_f^{\scriptscriptstyle V})_{kj}$$
 . Np

$$Nj$$
 7 \cdot v_k^{\perp} \cdot $(H_v^p)_{ik}$ \cdot k \cdot i (ATF: Acoustic

 C_k k k k p_k p

(scalar product) p

$$C_{k} = \frac{p_{k} \cdot p}{|p|}$$

$$= |p_{k}| \cos(\varphi_{k})$$

$$|p_{k}| \qquad k = 7$$
(3)

. Eq. 3

Eq. 2 Eq. 2

(complex

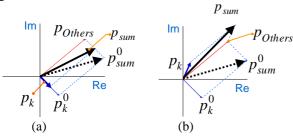
number)

 ϕ_k 7\? ϕ_k

Fig. 3. a

가

Fig.3 b



- (a) Reduction of amplitude and fixed phase
- (b) Reduction of amplitude and change of phase

Fig. 3 Resultant structure-borne noise according to change of amplitude $|p_k|$ and phase φ_k

2.2 (Design sensitivity analysis)

$$x_i$$
 7\text{ Eq. 2 } $\left(H_{\nu}^f\right)_k$ 7\text{ } p

$$\frac{\partial p_{k}(A, \varphi)}{\partial x_{i}} = \frac{\partial}{\partial x_{i}} \{ H_{v_{k}}^{p}(A, \varphi) \ v_{k}^{\perp}(A, \varphi) \}$$

$$\cong H_{v_{k}}^{p}(A, \varphi) \frac{\partial v_{k}^{\perp}(A, \varphi)}{\partial x_{i}}$$
(4)

, Eq. 4

$$\frac{\partial v_k^{\perp}(A, \varphi)}{\partial x_i}$$

- Forward difference technique

$$D_{i} = \frac{\partial p(\mathbf{x}^{0})}{\partial x_{i}} \cong \frac{p(\mathbf{x}^{0} + \Delta x_{i}) - p(\mathbf{x}^{0})}{\Delta x_{i}}$$
 (5)

- Central difference technique

$$\begin{vmatrix} p_k \end{vmatrix} \qquad D_i = \frac{\partial p(\mathbf{x}^0)}{\partial x_i} \cong \frac{p(\mathbf{x}^0 + \Delta x_i) - p(\mathbf{x}^0 - \Delta x_i)}{2 \Delta x_i}$$

$$(6)$$

 Δx_i x_i

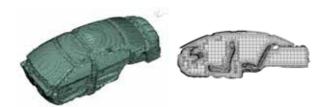
(finite increment) .

3.

3.1

200Hz (low frequency) 200Hz 800Hz (medium frequency) Fig. 4.a Fig. 4.b (BIW) (trim) (floor carpet), (seat), (cockpit), (console) (trunk door) (passenger door)

(a) Structural model including front suspension system

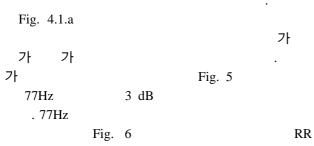


Iso-view Section view
(b) Acoustic cavity model: Outer and section view
Fig. 4 Finite element models for vibro-ac

Fig. 4 Finite element models for vibro-acoustic coupling analysis of a passenger vehicle

(IPI), (VTF) (NTF)

pressure peak) .



WHEL, RRFLR, ROOF, DASH, PSHLF 가

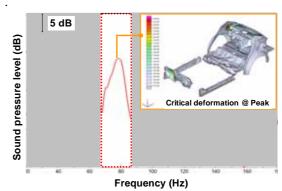


Fig. 5 Critical NTF(Noise Transfer Function) and concerned deformation shape with respect to the peak@77Hz

3.2

(sound

Fig. 6 . Fig. 6 Eq. 3
$$7 + Fig. 5 Fig. 7$$
 RRWHEL DASH $7 + Fig. 7$
$$(p_{sum})$$

 $p_{\mathit{RRWHEL}} > p_{\mathit{RRFLR}} > p_{\mathit{ROOF}} > p_{\mathit{DASH}} > p_{\mathit{PSHLF}} > \cdots$

RRFLR ROOF 가 DASH

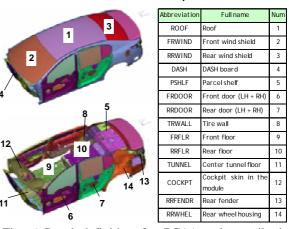
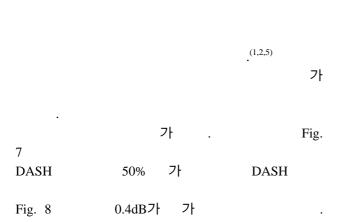


Fig. 6 Panel definition for PCA(panel contribution analysis)



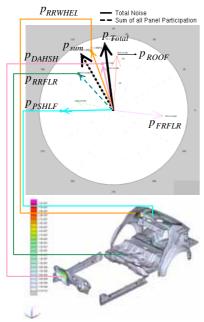
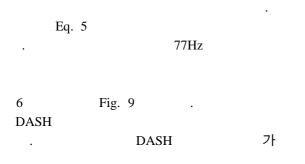
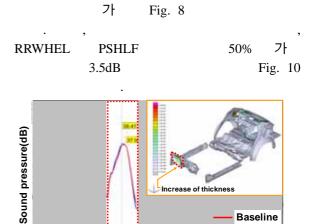


Fig. 7 Critical deformation shape vs. PCA(Panel Contribution Analysis) with respect to the critical NTF(Noise Transfer Function) at 77Hz

3.3





Baseline

Modified

Fig. 8 Comparison of NTF(Noise Transfer Function) according to thickness increase of DASH panel which has positive panel sensitivity with respect to the critical NTF(Noise Transfer Function) at 77Hz

Frequency (Hz)

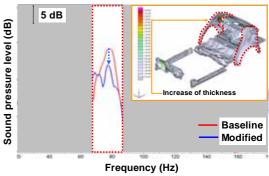


Fig. 10 Comparison of NTF(Noise Transfer Function) according to thickness increase of panels which have negative design sensitivity as well as positive panel sensitivity, with respect to the critical NTF(Noise Transfer Function) at 77Hz

4.

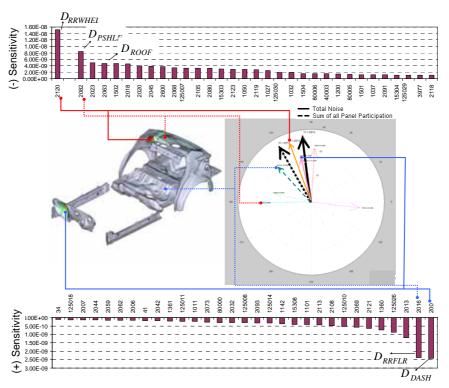
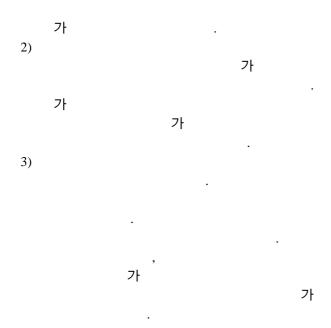


Fig. 9 Comparison of PCA(Panel Contribution Analysis) and design sensitivity of the critical NTF(Noise Transfer Function) at 77Hz with respect to according to panel thickness



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